## MATH 854: FLUID MECHANICS III

STREAM: Y1 S2
DAY: WEDNESDAY, 2:00 - 5:00 P.M.
TIME: 3 HOURS

DATE: 03/05/2023 INSTRUCTIONS

1. Do not write anything on this question paper.
2. Answer Question ONE Compulsory and any other TWO (2) Questions.

## QUESTION ONE (30 MARKS)

a) Describe briefly the term dimensional homogeinity.
b) Find an expression for the drag force on smooth sphere of diameter $D$, moving with uniform velocity $V$ in a fluid density $\rho$ and dynamic viscosity $\mu$.
(7marks)
c) State the Buckingham's $\pi$-theorem.
d) The resistance $R$ experienced by a partially submerged body depends upon the velocity $V$, length $l$, viscosity of the fluid $\mu$, density of the fluid $\rho$ and gravitational acceleration $g$. Obtain a dimensionless expression for $R$.
e) Briefly explain the following concepts in terms of Mach number:
i) Subsonic flow
ii) Sonic flow
iii) Supersonic flow
(3marks)
f) Air at a pressure of $240 \mathrm{kN} / \mathrm{m}^{2}$ and temperature $30^{\circ} \mathrm{C}$ is moving at a velocity of $200 \mathrm{~m} / \mathrm{s}$. Calculate the stagnation pressure if:
i) Compressibility is neglected.
ii) Compressibility is accounted for.

Take $R=287287 \mathrm{~J} / \mathrm{kgK}, \gamma=1.4$
(8marks)

## QUESTION TWO ( 15 MARKS)

a) Derive the differential equation of a perfect gas given by the equation $\frac{d p}{p}-\frac{d \rho}{\rho}-\frac{d T}{T}=0$
(3marks)
b) Briefly explain the THREE basic thermodynamic processes involved in compressible fluid flow.
(6marks)
c) A gas is flowing through a horizontal pipe. On a section where crosssection area is $100 \mathrm{~cm}^{2}$, the pressure and temperature are found to be 4 bar (gauge) and $40^{\circ} \mathrm{C}$ respectively. At another section where the area of the cross-section is $50 \mathrm{~cm}^{2}$ the pressure is recorded 3 bar (gauge). If the mass rate of flow of gas through the pipe is $0.6 \mathrm{~kg} / \mathrm{s}$, find the velocities of the gas at these sections, assuming an isothermal change. Take $R=287 \mathrm{~J} / \mathrm{kgK}$ and atmospheric pressure $=1$ bar.
(6marks)

## QUESTION THREE (15 MARKS)

a) A gas with a velocity of $400 \mathrm{~m} / \mathrm{s}$ is flowing through a horizontal pipe at a section where pressure is $80 \mathrm{kN} / \mathrm{m}^{2}$ absolute and temperature $50^{\circ} \mathrm{C}$. The pipe changes in diameter and at this section, the pressure is $120 \mathrm{kN} / \mathrm{m}^{2}$ absolute. Find the velocity of the gas at this section if the flow of the gas is adiabatic. Take $R=287 \mathrm{~J} / \mathrm{kgK}$ and $\gamma=1.4$.
(10marks)
b) An aeroplane is flying at a height of 15 km where temperature is $-40^{\circ} \mathrm{C}$. The speed of the plane is corresponding to $M=2$. Find the speed of the plane if $R=287 \mathrm{~J} / \mathrm{kgK}$ and $\gamma=1.4$
(5marks)

## QUESTION FOUR (15 MARKS)

a) A supersonic aircraft flies at an altitude of 3.6 km where temperature is $4^{0} \mathrm{C}$. Determine the speed of the aircraft if its sound is heard 4 seconds after its passage over the head of an observer. Take $R=287 J / \mathrm{kgK}$ and $\gamma=$ 1.4
(7marks)
b) An aeroplane is flying at $900 \mathrm{~km} / \mathrm{hr}$ through still air having a pressure of $78.5 \mathrm{kN} / \mathrm{m}^{2}$ (absolute) and temperature $-10^{\circ} \mathrm{C}$. Calculate on the stagnation point on the nose of the plane:
i) Stagnation pressure
ii) Stagnation temperature
iii) Stagnation density

Take $R=287 \mathrm{~J} / \mathrm{kgK}$ and $\gamma=1.4$

